

Presented by

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SIM Key Objectives

Science *

Technology

Demonstrate Technology of Synthesis Imaging

Demonstrate Technology of Starlight Nulling

Usher in the Era of
Long Baseline, Short Wavelength
Interferometry for
Astrophysical Observation

Indirect Planet Detection

Down to a Few Earth Masses

(goal: 1 μas; min: 3 μas)

* Technology maturation over the next few years will determine the ultimate achievable performance

Ultra Precision Global Astrometry (goal: 4 µas;

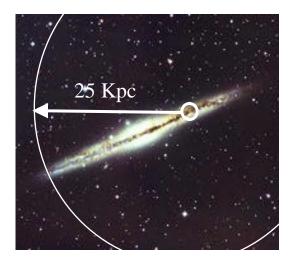
min: better than 30µas)

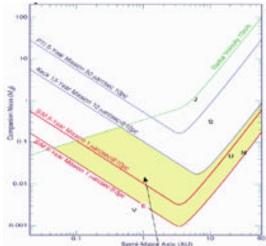
Space Interferometry Mission

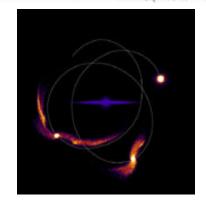
Astrophysics with SIM

New Discoveries ... and Answers to Age-Old Questions

- Calibration of the Cosmic Distance Scale
 - Cepheids and nearby spiral galaxies
- Dynamics of the galaxy
- Fundamental stellar astrophysics
- Star/Planetary system formation
 - The quest for terrestrial planets
- Masses of black hole and neutron star binaries
 - Probe nature of dark matter via gravitational lensing
 - Dynamics of the local group of galaxies, dark matter in nearby galaxies, and between galaxies
 - Frame tie between SIM (optical) and ICRF (radio)



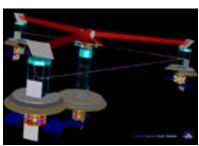




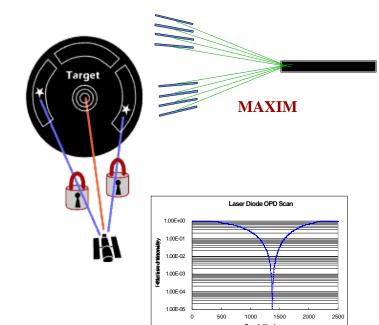
SIM as a Technology Precursor to Future Missions

- SIM is an integral part of the flow of technology within the Origins Program and the Space Science Enterprise
 - TPF and future PlanetImaging Interferometers
 - Long baseline Interferometers from submm to X-rays
- SIM is a <u>unique</u> precursor for TPF in the following areas:
 - Angle and path length feed forward in space
 - Space demonstration of nulling to 10⁻⁴ (vis)
 - Imaging with a nulling interferometer (rotate a baseline while maintaining a deep null, moving delay line)





TPF SPECS



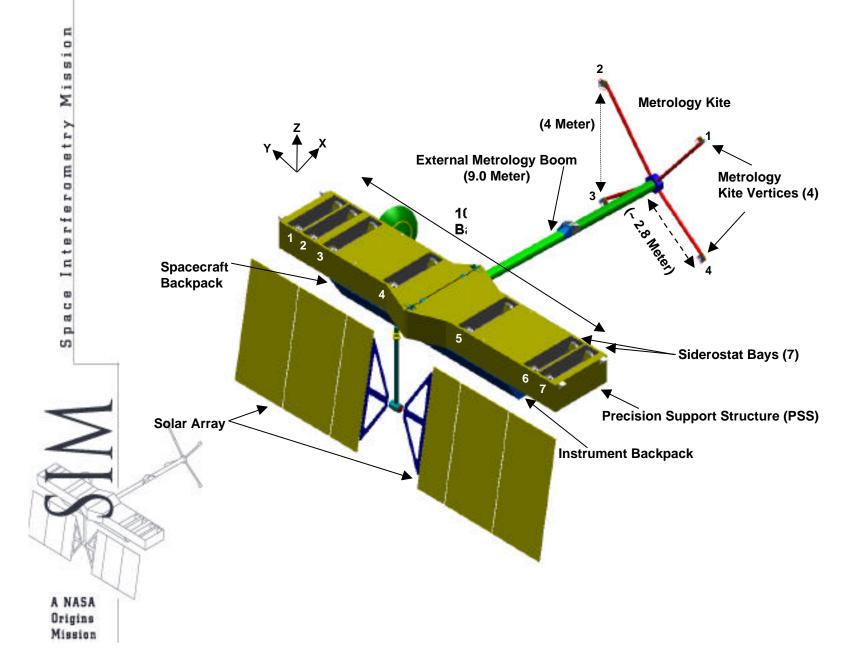
Optical Nulling

Space Interferometry Mission

- 3 collinear
 Michelson Stellar
 Interferometers
- 10 meter baseline
- Visible wavelength
- EELV (Atlas V 421) launch vehicle
- Earth-trailing solar orbit
- 5-year mission life
 - SIM is a JPL,
 Caltech, Lockheed
 Martin, and
 TRW partnership



SIM Flight System Configuration



The SIM Partnership: Four Partners

One Team



Interferometer I&T Interferometer Operations



Spacecraft
Precision Structure

Assembly, Test, & Launch Operations

S/C Operations





Interferometry Science Center Science Data Analysis and Archiving Science Operations Science Planning Science Community Interface Outreach

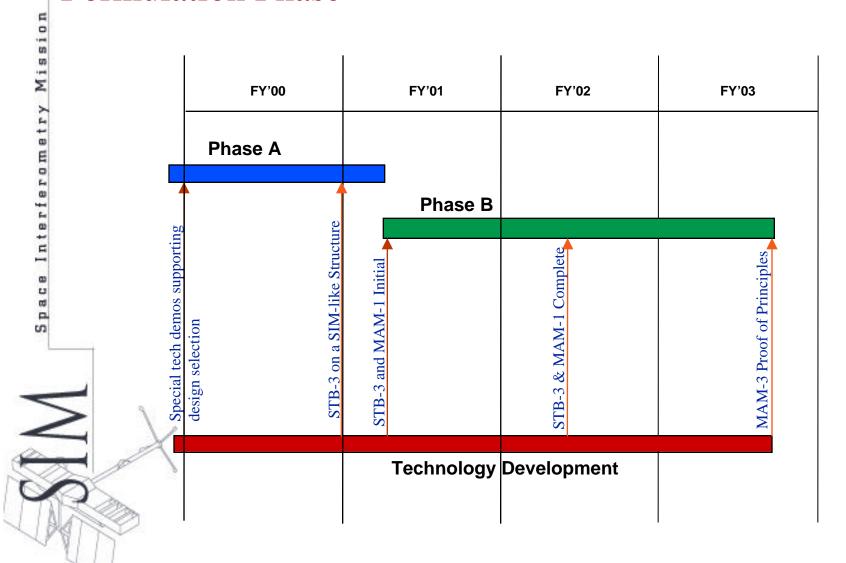




Project Management System Engineering Integrated Modeling Real Time Control Mission Systems Mission Assurance Risk Management



Near-Term Technology Schedule Formulation Phase



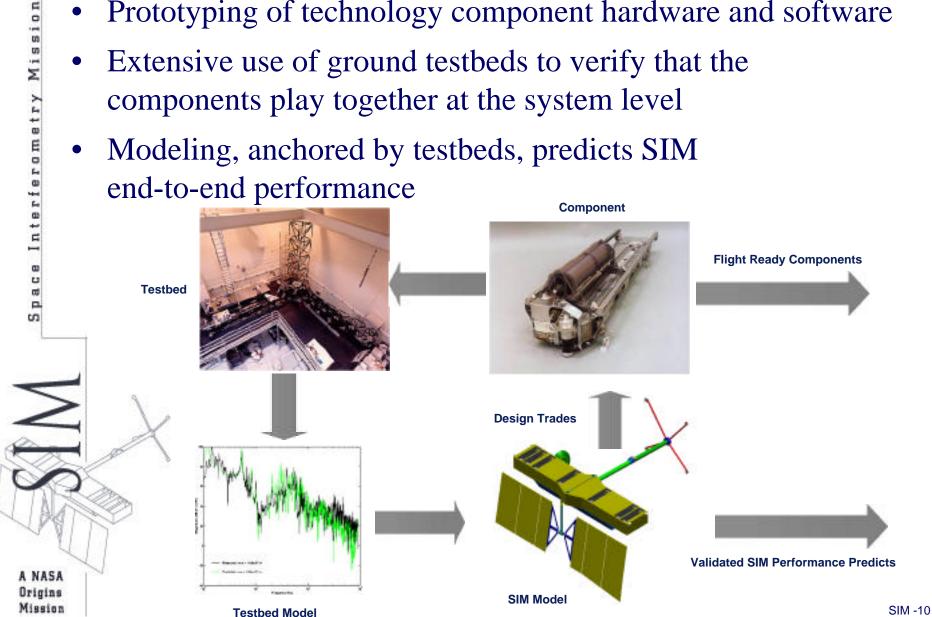
Required Technologies

- Knowledge of positions of optical elements to 25 Picometers (100 pm = diameter of a hydrogen atom)
 - Picometer laser metrology
 - Picometer starlight fringe position measurement
- Nanometer control of optical path difference (75,000 nm = thickness of a human hair)
 - Nano-g vibration isolation
 - Nanometer/nanoradian active optics
 - Micron stability of structures
- Picometer thermal deformation of optics
 - Millikelvin thermal control of optics
 - 10,000:1 starlight nulling beam combination in the visible

Approach to Technology Development

- Prototyping of technology component hardware and software
- Extensive use of ground testbeds to verify that the components play together at the system level

Modeling, anchored by testbeds, predicts SIM end-to-end performance



SIM System Level Testbeds

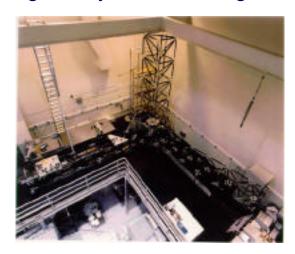
• SIM is developing four system level testbeds to demonstrate technology performance at the system level

- Two near full-scale testbeds that demonstrate nanometer control in air
 - System TestBed 1 (STB-1) is a one-baseline testbed
 - System TestBed 3 (STB-3) is a three-baseline testbed
 - Two sub-scale testbeds that demonstrate picometer knowledge in vacuum
 - MAM-1 is a 1/5 scale one-baseline testbed
 - MAM-3 is a 1/3 scale three baseline testbed



SIM System Testbeds (STB-1 & STB-3)

- STB-1 has been in operation since 1994
- Demonstrating nanometer stabilization for a single interferometer on a flexable structure
- Validating SIM dynamic modeling accuracy



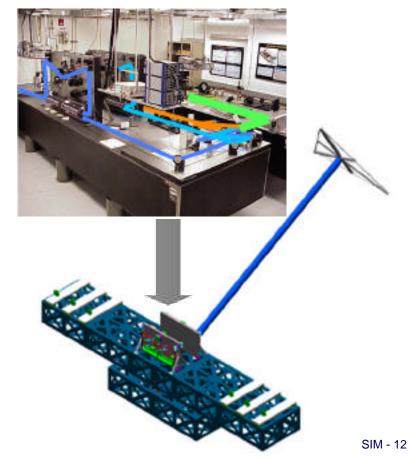
Dim Star

100.0

Bright Star

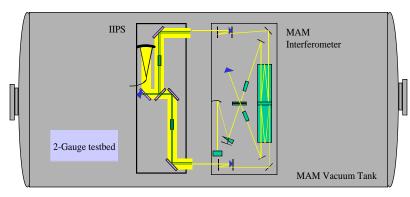
Astrometric Requirement Nulling Requirement

- STB-3 is presently operating on optical tables
- Will demonstrate pathlength feed forward, dim star control capability
- Will transition to a flexable structure in December 2000



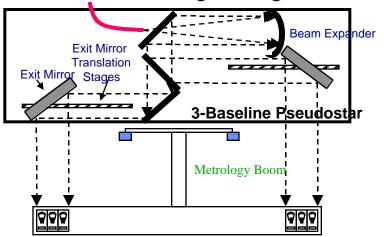
Micro-Arcsecond Metrology Testbeds (MAM-1 & MAM-3)

- MAM-1 is a 1/5 scale, one interferometer testbed
- Will demonstrate that metrology and starlight sensing can be integrated and provide consistent outputs at the picometer level
- Development of the interferometer and pseudo-star are in process
- Functional testing will take place in 2001



MAM-1 schematic

- MAM-3 is a 1/3 scale, three interferometer testbed
- Will demonstrate the transfer of guide star position knowledge with the precision required to measure science star positions in inertial space
- Requirements definition and conceptual design in process
- Will complete in time to influence the flight design



MAM-3 schematic

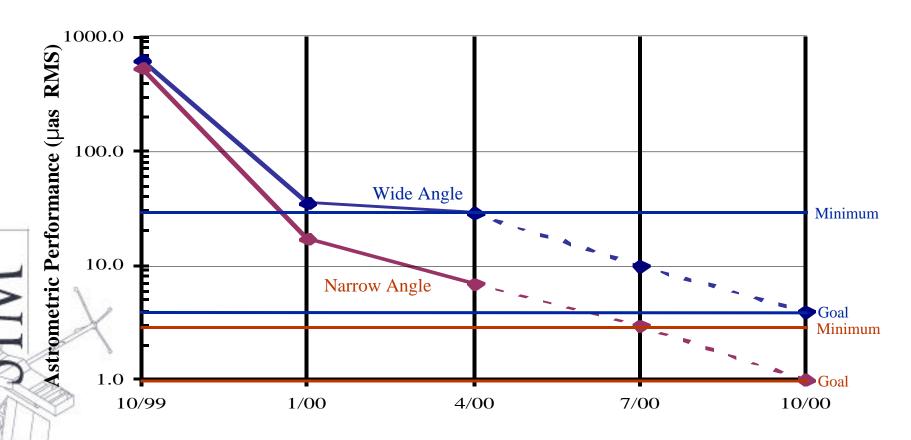
SIM Performance Metric

What If We Froze the Technology Now

				SIM Wide angle (WA) / Narrow Angle (NA)
	Parameters	Parameter Performance Today	Parameter Performance Needed by NAR	What if all other parameters improve to the NAR level but no more improvement on this one
1	Beam launcher thermal sensitivity (bulk, gradient)	100 pm/mk, 2500 pm/ml	2 pm/mk, 50 pm/mk	28 μas/ 4.8 μas •
2	Cyclic averaging residual error per gauge	100 pm	5 pm	4 μas/ 3 μas
3	Pointing deter error per gauge	750 mas	75 mas	5 μas/ 2.8μas
4	Corner cube surface quality	lamda/100	lamda/500	5 μas/ 1.2μas
5	Wide angle error due to beam diffraction	1000 pm	100 pm	8.5 µas (WA)
6	Narrow angle error due to beam diffraction	67 pm	2.7 pm	1.7 μas (NA)
7	Wide angle error due to polarization effects on corner cubes	375 pm	15 pm	10 μas (WA)
8	Narrow angle error due to polarization effects on corner cubes	25 pm	1 pm	2 μas (NA)
9	1-D absolute metrology accuracy	30 μm	3 µm	4 μas/1.2μas
10	Wide angle PSS end-to-end thermal deformation	100 µm	10 μm	4.4 μas (WA)
11	Narrow angle PSS end-to-end thermal deformation	10 μm	1 μm	0.9 μas (NA)
12	Systematic fringe measurement error	250 pm	30 pm	5.1 μas/3 μas •
	Wide Angle Performance (General Astrophysics) 4 µas (goal); 30 µas (min)	29 µas	3.9 µas	
	Narrow Angle Performance (Planet Detection) 1 µas (goal); 3 µas (min)	7 μas	0.8 µas	SII

SIM Performance Metric

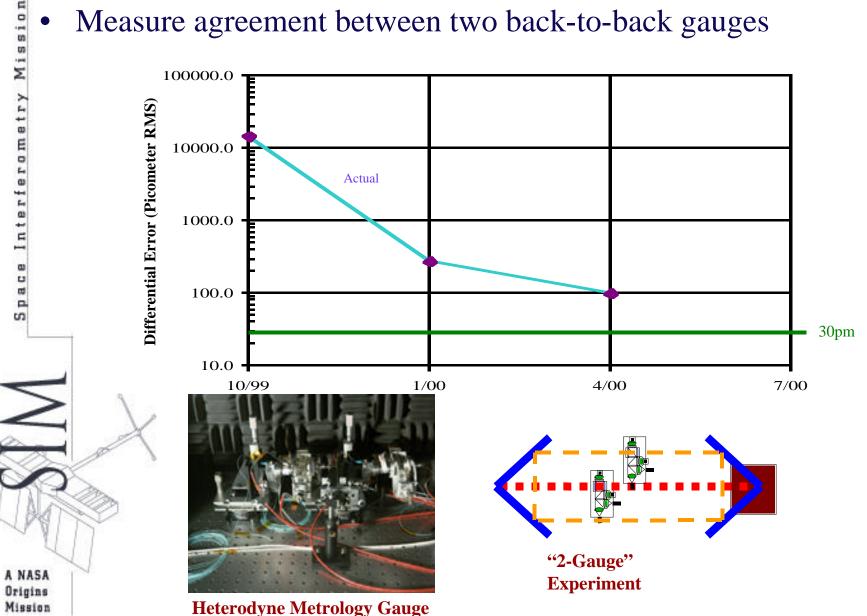
- Calculated Wide Angle based on Component Performance
- Calculated Narrow Angle based on Component Performance



Space Interferometry Mission

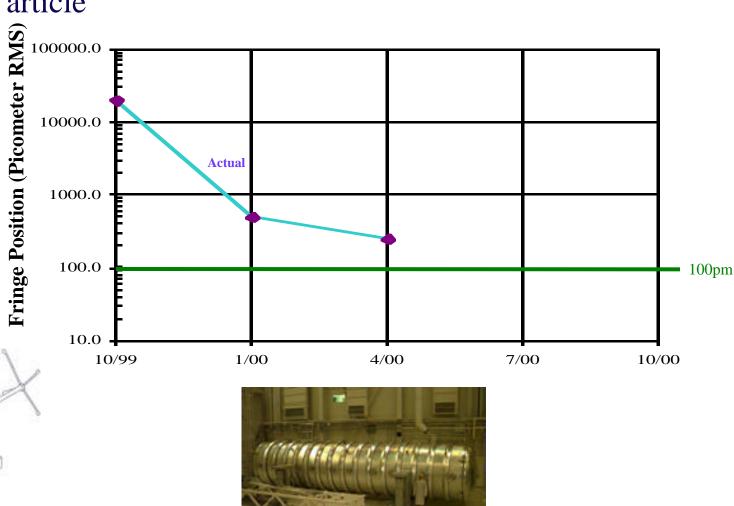
Metrology Gauge Experiment – Progress

Measure agreement between two back-to-back gauges



White Light Experiment – Progress

• "Stellar" fringe measurement accuracy – MAM interferometer test article



A NASA Origins Mission

Mission

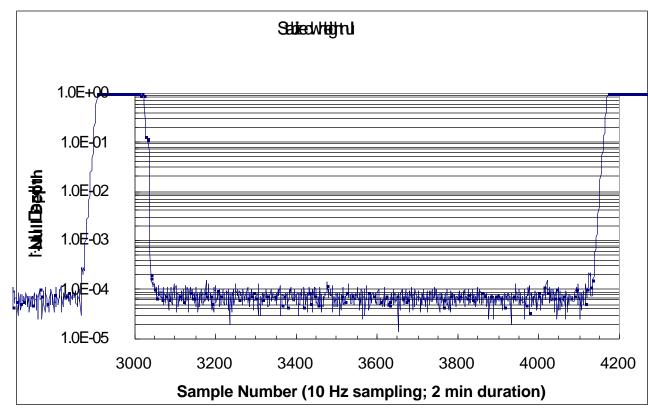
MAM Testbed

Stabilized White Light Null

- Two years ago this was considered the most difficult technology for SIM
- This year we have nearly achieved the requirement
 - White light null
 - Stable at less than 10⁻⁴
 - 18% bandwidth

A NASA

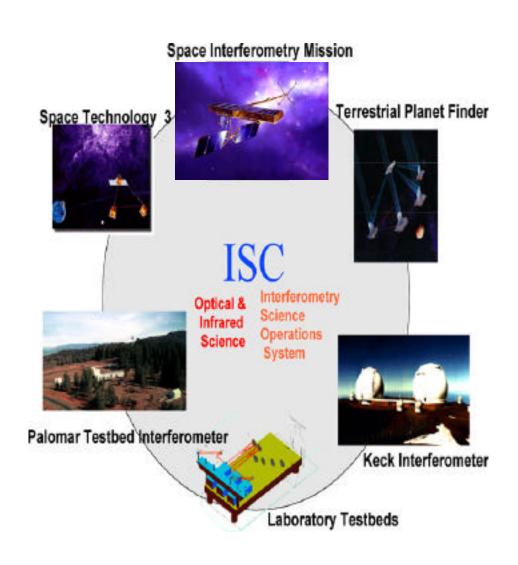
• Single polarization



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Interferometry Science Center

- Dr. Anneila Sargent appointed ISC Director
- JPL/Caltech collaboration
- Multi-mission capability for interferometry
 - Built upon Infrared Processing and Analysis Center heritage



Science AO

- The SIM Science AO was released on February 25, 2000
- Proposals due May 26, 2000
- Announcement of selections August 2000
- Selections will be for the following:
 - As many as 8 Key Project Science Investigations
 - 1 Education and Public Outreach (E/PO) Scientist
 - Minimum of 2 Data Scientists
 - Minimum of 2 Instrument Scientists
 - Maximum of 2 Interdisciplinary Scientists
 - 1 Imaging and Nulling Scientist
- This AO will distribute about 60% of the total observing time for SIM
 - Up to 10% for each Key Project Science Investigation
 - Up to 1% for each of the Discipline Scientists
 - 23 Notice of Intents were received for Key Project Science
- 27 Notice of Intents were received for individual scientists

Last Year's Accomplishments – Technology

- Significant progress in nanometer technologies
 - Continued improvements on STB-1
 - Obtained STB-3 first fringes
- Nulling technology demonstrated at required level
 - Achieved 10⁻⁴ stable white light null
- Significant progress has been made in picometer technology, and considerable momentum established
 - Achieved 100pm precision with met gauge (Hydrogen atom diameter)
 - Demonstrated 250pm white light fringe measurement capability
- Successful at modeling nanometer performance of a full scale interferometer on a flexible structure
- Demonstrated ability to measure and predict millikelvin temperature response of optics
 - Demonstrated ability to measure optical surface deformation at the 30pm level

Last Year's Accomplishments – Flight System

- Completed instrument design selection
 - SIM Classic is our reference design
- Developed a design description document for the reference design
- Development of level 1, 2, 3, and 4 requirements are on schedule for our September System Requirements Review (SRR)
- Began the SIM Independent Assessment
 - Held first Technical Interchange Meeting April 4 & 5

Coming Next Year

Technology

- Demonstrate dim star fringe tracking on STB-3 on a flexible structure
- Obtain first fringes on MAM-1
- Demonstrate SIM performance levels for the Met Gauge (≤ 30pm) and the white light fringe (≤ 100pm) measurement capability
- Obtain initial picometer deformation measurements of optics under millikelvin temperature changes

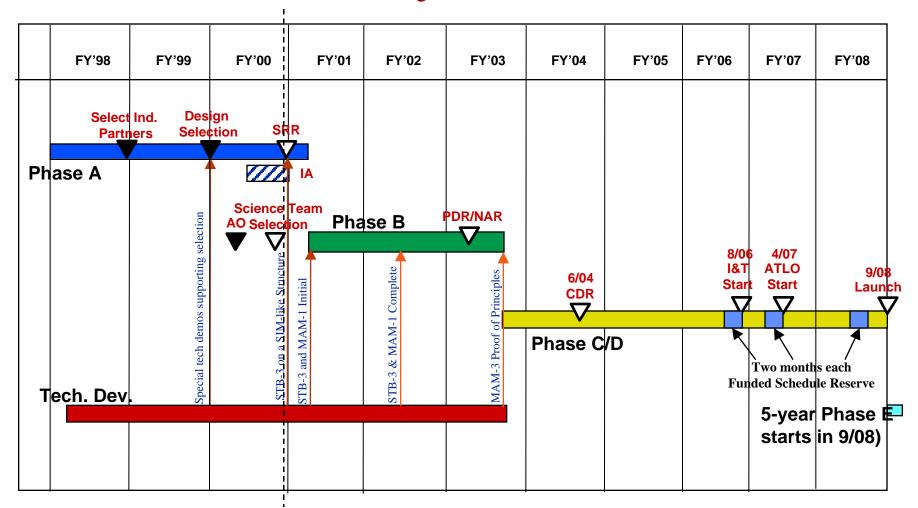
Flight System

- Complete the System Requirements Review (October 2000)
- Complete the Independent Assessment (October 2000)
- Ready to transition into Phase B (December 2000)

SIM Schedule Update

- Previous Formulation Phase schedule margin was inadequate
 - Completion of MAM-1 and STB-3 by PDR/NAR was very success oriented
 - Completion of MAM-3 proof of concept demonstration in time to influence the flight design was likely not achievable
 - Completion of critical brassboards by Implementation Phase start was success-oriented
- Previous 4 1/4 year Implementation Phase schedule inadequate
 - Delivery of metrology components on schedule was success oriented
 - Interferometer I&T and ATLO schedule was success oriented
- Current Schedule significantly reduces Project risk by extending the Formulation Phase 1 1/4 years and Implementation Phase 1 year
 - Provides adequate schedule for completion of MAM-1 and STB-3
 - Provides adequate schedule to complete MAM-3 proof of concept prior to Implementation Phase start
 - Provides adequate schedule for critical brassboard development
 - Provides the needed additional 6 months to the metrology component schedule
 - Provides the needed additional 6 months to the I&T/ATLO schedule

SIM Project Schedule



Changes since last year:

- Phase B extended 1 1/4 years
- Phase C/D extended 1 year
- Launch moves out 2 1/4 years

Program Resiliency

- Realizability of a complex mission demands program resiliency as measured by its various programmatic and technical margins
- For SIM, these include
 - Mass margin 37%
 - EELV (Atlas V 421) selected as the Launch Vehicle
 - − Power margin − 25%
 - Budget reserve 30% (Phase C/D)
 - Schedule reserve 6 months (Phase C/D)
 - Science Science floor has been defined
 - Launch opportunities Not constrained

Outreach Highlights – Michelson Fellowship Program

A program to support the scientific community in building expertise in optical and IR interferometry



Interdisciplinary team of undergraduate students designing beamcombiner for Keck Interferometer

Five integrated components

- Post-doctoral Fellows
- Graduate Student Fellowships
- Undergraduate Research
- Interferometry Summer School
- Distinguished Lecture Series

Status and Plans

- Five postdoctoral and seven graduate student fellowships awarded (among others at MIT, Harvard, U of A, SUNY, University of New Mexico, Georgia State, Caltech)
- Sponsored a Harvey Mudd Undergraduate Clinic in collaboration with Keck Interferometer
- The first summer school in Pasadena was very successful. Second will be held August 2000 at UC Berkeley

A NASA Origins Mission

Space Interferometry Mission

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Takeaway Chart

Programmatic

- Science AO is out and the Science Team will be on board this fiscal year
- SIM is doable for the new cost estimate
- The current schedule significantly reduces Project risk
- The Project will be ready for transition to Phase B this Fall

Technology

- Nulling technology is in hand
- Nanometer technology is nearly in hand
- Significant progress in picometer technology, but we still have a ways to go with picometer system level testbeds

Flight System

- Independent Assessment is underway
- Development of system requirements are on track for the System Requirements Review in September 2000

Space Interferometry Mission

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For More Information on SIM, See Our Website at:

http://sim.jpl.nasa.gov

